



Advancements in Coupled Flow and Material Modeling for Entry Systems



14th Ablation Workshop

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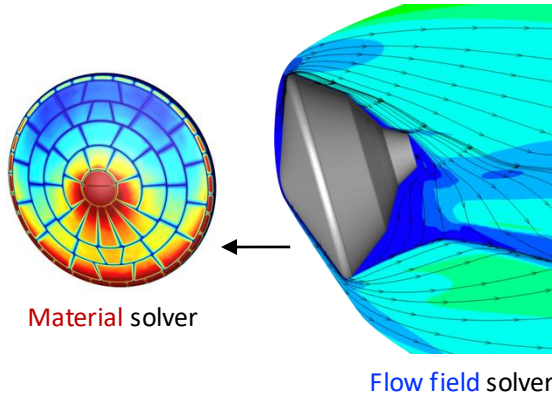
³Oak Ridge Associated Universities

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Main Strategies to Couple Flow and Material

One-Way Coupling

Exchange of boundary conditions from flow solver to material solver



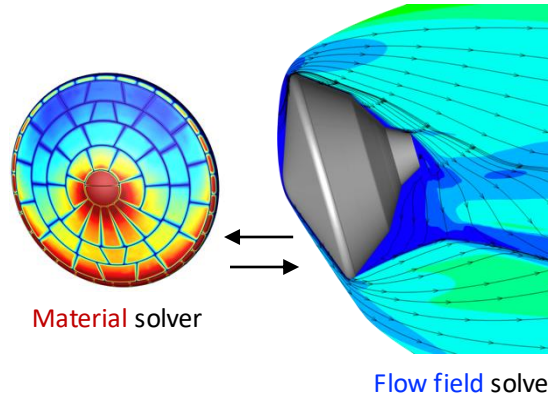
Material solver

Flow field solver

- decouples the **flow field** entirely from the **material** response;
- assumes that the **flow field** reaches steady state;
- assumes that the **material** ablation is in chemical equilibrium.

Two-Way Coupling

Exchange of boundary conditions between flow and material solvers



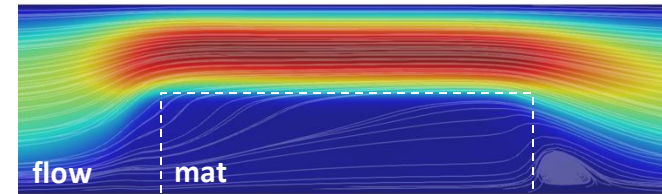
Material solver

Flow field solver

- is based on a iterative procedure to couple the **flow field** and **material**;
- assumes that the **flow field** adapts instantaneously to the **material** shape change.

Unified Coupling

No exchange of boundary conditions between flow and material



Combined **flow field** and **material** solver

- solves the **flow field** and the **material** in the same computational domain;
- models the interface progression as a cause of the **material** degradation;
- requires effective properties at the interface region.



Multi-Physics One-Way Coupling for Venus Mission Concept

Aerosol Capture
AERACEPT^[1-3]

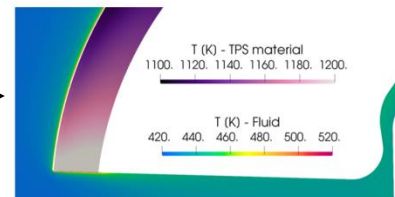
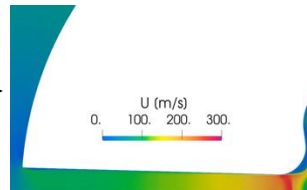
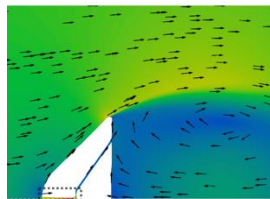
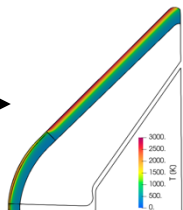
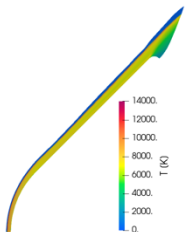
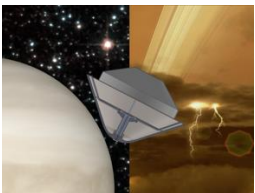
Hypersonic CFD
DPLR^[4]

Material Response
PATO^[5]

Subsonic CFD
OpenFOAM^[6]

Particle Tracking
OpenFOAM

Thermal Analysis
OpenFOAM



Hypersonic CFD to Material

- Heat transfer coefficient C_H [kg/m²/s]
- Recovery enthalpy h_r [J/kg]
- Wall pressure p_w [Pa]

Surface Mass & Energy Balance

$$B'_c = \frac{B'_g(z_{C,g} - z_{C,w}) + z_{C,e} - z_{C,w}}{z_{C,w} - 1}$$

$$-\left(\bar{\mathbf{k}}_w \cdot \frac{\partial T_w}{\partial \mathbf{n}}\right) = C'_H(h_r - h_w) \cdot$$

$$+ \dot{m}_g(h_g - h_w) + \dot{m}_c(h_c - h_w)$$

$$- \varepsilon_w \sigma (T_w^4 - T_\infty^4) + \alpha_w q_{rad}$$

Trajectory/TPS to CFD/Particle

- Freestream conditions:
 U_∞ [m/s], p_∞ [Pa], T_∞ [K]
- TPS wall temperature: T_w [K]

Particle Motion & Heat Transfer

$$m_p \partial_t \mathbf{v}_p = \sum_i \mathbf{F}_i$$

$$\mathbf{F}_D = \frac{3}{4} C_D \frac{m_p \mu_g \epsilon_g^{-1.65}}{\rho_p (d_p)^2} (\mathbf{v}_g - \mathbf{v}_p)$$

$$\partial_t T_p = \frac{S_p}{m_p c_{p,p}} (\mathcal{H}_g (T_g - T_p) + F \sigma \varepsilon_w (T_w^4 - T_p^4))$$

Aerosol Capture Predictions

Sample Quantification

- 623 million droplets, mostly 2 μm in diameter
- Total volume of 24.95 μL , mostly 8 μm in diameter
- Capture efficiency below 1% for small particles (0.4 μm and less)

Sample Integrity

- Limited effect from convection: maximum increase of 13 K
- Significant effect from radiation: 10% reduction in total volume

[4] Wright, M.J., et al. "Data Parallel Line Relaxation (DPLR) Code User Manual: Acadia-Version 4.01.1." 2009.

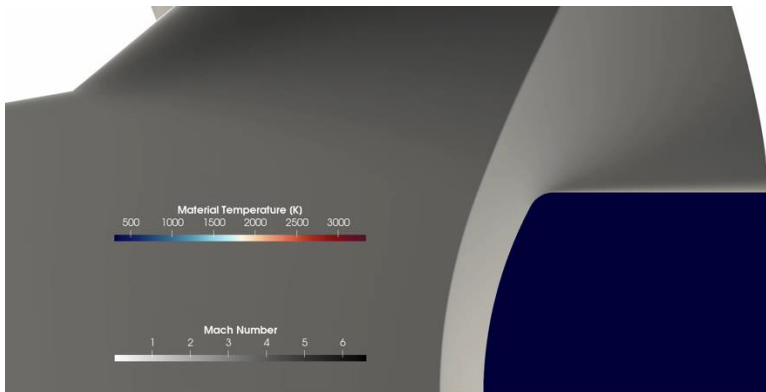
[5] Lachaud, J., et al. "A generic local thermal equilibrium model for porous reactive materials submitted to high temperatures." 1977.

[6] Weller, H.G., et al. "A tensorial approach to computational continuum mechanics using object-oriented techniques." 1998.

Hypersonic CFD
US3D^[7]

Coupler
Ares^[8]

Material Response
Icarus^[9]



Arc-Jet Test for Model Validation

- IHF-385 IsoQ sample with 3MDCP thermal protection material
- Exposure times = 38 - 39 sec
- Recession measured = 4.3 - 4.8 mm
- Recession computed = 4.9 mm
- Validate 3MDCP material model



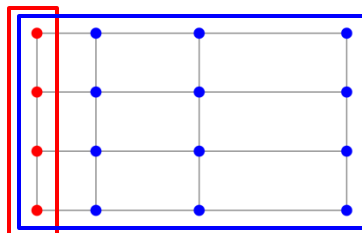
Two-way Coupling with Mesh Motion

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algo
Load MR data into CFD code
Set coupling BC, frequency and algorithm type
Solve CFD to quasi-steady state
While MR is running:
  Solve MR + Update MR mesh
  If coupling frequency reached:
    Move CFD mesh based on MR mesh boundary
    Solve CFD to quasi-steady state + Update MR BC
  
```

Mesh Motion Approach

$$\mathbf{M}^{bc} = \phi(\|\mathbf{x}_i^{bc} - \mathbf{x}_j^{bc}\|)$$



$$\mathbf{A}^{in} = \phi(\|\mathbf{x}_i^{in} - \mathbf{x}_j^{bc}\|)$$

\mathbf{M}^{bc} and \mathbf{A}^{in} are built using Wendland^[10]
Radial Basic Function: $\phi(\|\mathbf{x}\|) = (1 - \|\mathbf{x}\|)^2$

$$\mathbf{r}_{MR}^{bc} = \mathbf{M}_{MR}^{bc} \boldsymbol{\alpha}$$

$$\boldsymbol{\alpha} = (\mathbf{M}_{MR}^{bc})^{-1} \mathbf{r}_{MR}^{bc}$$

$$\mathbf{r}_{MR}^{in} = \mathbf{A}_{MR}^{in} \boldsymbol{\alpha}$$

$$\mathbf{r}_{CFD}^{in} = \mathbf{A}_{CFD}^{in} \boldsymbol{\alpha}$$

$\boldsymbol{\alpha}$ is computed once on MR grid, then re-used on CFD grid, significantly reducing computational cost.

[10] Rendall, T., et al. "Efficient mesh motion using radial basis functions with data reduction algorithms." 2009.



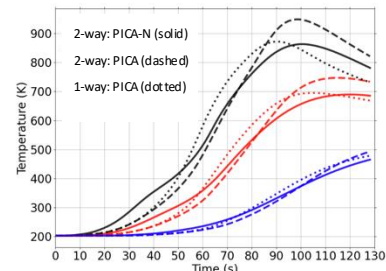
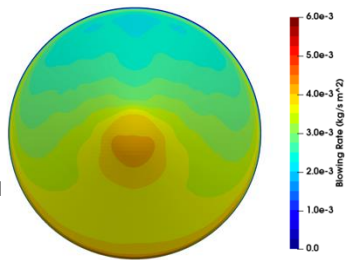
Approaches for Two-Way Coupling with Pyrolysis Gas Blowing

Hypersonic CFD
DPLR & US3D

Coupler
Ares

Radiation
NEQAIR^[11] & NERO^[12]

Material Response
PATO & Icarus



credit: John M. Thornton (2024 ARC-TSM contractor)^[13]

Two-way Coupling with Pyrolysis Gas Blowing

```

For all trajectory points:
  Solve CFD without blowing BC + solve Radiation
  Update MR BC using CFD results [q_w, p_w]
  Solve MR with blowing correction
  While |T_{w,CFD} - T_{w,MR}| > threshold:
    Update CFD BC using MR/GSI [m_g, y_w, m_c, rho_g]
    Solve CFD with blowing BC + solve Radiation
    Update MR BC using CFD results [q_w, p_w]
    Solve MR without blowing correction
  
```

algo

Surface Mass & Energy Balance Approach

Wall Composition using B'

$$z_{w,k} = \frac{z_{e,k} + B'_g z_{g,k}}{\sum_k^{N_e} z_{ws,g,k}}$$

$$z_{w,C} = \frac{L + z_{e,C} + B'_g z_{g,C}}{\sum_k^{N_e} z_{ws,g,k}}$$

$$L = \max(100B'_g, 200)$$

Equilibrium $(p_w, T_w, z_{w,k}) \rightarrow x_{wg,k}$

$$h_w = \sum_i^{N_s} x_{wg,i} h_{wg,i}$$

Blowing Correction

$$C_H = \frac{q_w}{h_e - h_w}$$

$$\frac{C'_H}{C_H} = \frac{\ln(1 + 2\lambda(B'_g + B'_c))}{2\lambda(B'_g + B'_c)}$$

Limitations

After the peak heating,
 $h_w - h_e < 0$

$$q_w = q_{conv} + q_{diff} + q_{adv}$$

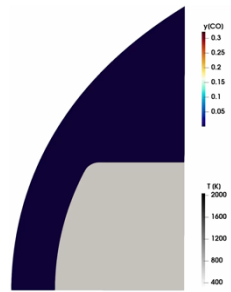
Diffusive Flux Approach^[14-16]

Wall Composition using J_k

$$J_k + (\dot{m}_g + \dot{m}_c)z_{w,k}$$

$$= \dot{m}_g z_{g,k} + \dot{m}_c z_{c,k}$$

Computing the mass balance using diffusive flux from CFD allows the computation of \dot{m}_c without B' assumption



Two-way coupling with pyrolysis gas blowing (19 species) and mesh motion for simulating a PICA arc-jet case based on Milos^[17]

[17] Milos, F.S. and Chen, Y.-K. "Ablation and thermal response property model validation for phenolic impregnated carbon ablator." 2010.

Unified Solver: Governing Equations^[18]

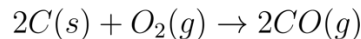
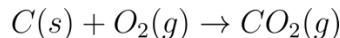
$$\partial_t(\epsilon_g \mathbf{U}) + \partial_x \cdot [\epsilon_g (\mathbf{F}^{inv} + \mathbf{F}^{vis})] = \mathbf{S}$$

$$\partial_t \epsilon_s = \frac{-\dot{\omega}^{het}}{\rho_s}$$

$$\mathbf{S} = \begin{bmatrix} \epsilon_g \dot{\omega}^{chem} + \dot{\omega}^{het} \\ -\mathbf{F}^{drag} \\ -\mathbf{F}^{drag} \cdot \mathbf{u} \end{bmatrix}$$

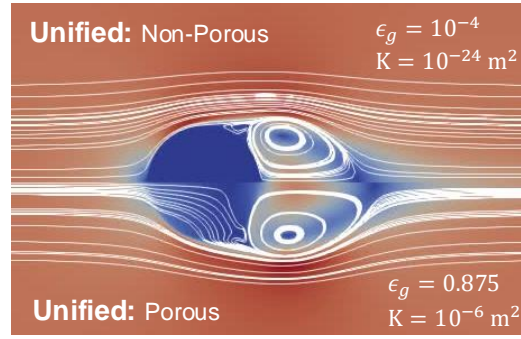
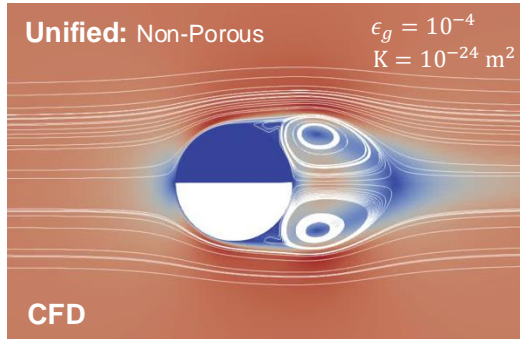
$$\mathbf{F}^{drag} = -\mu \bar{\mathbf{K}}^{-1} (\bar{\mathbf{I}} + \bar{\mathbf{F}}) \epsilon_g^2 \mathbf{u}$$

Unified Solver: Oxidation Model^[19]

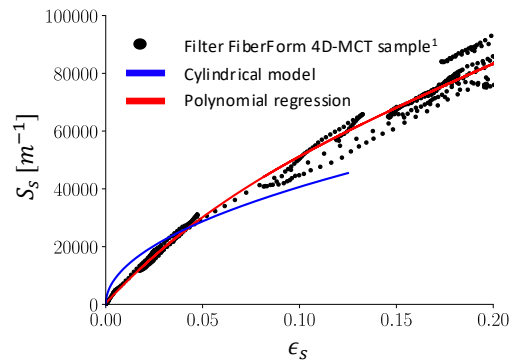


$$\dot{\omega}_i^{het} = -S_f k_i \rho_{O_2} \frac{M_i}{M_{O_2}}$$

$$\dot{\omega}^{het} = \sum_i \dot{\omega}_i^{het} \quad i \in [O_2, CO, CO_2]$$

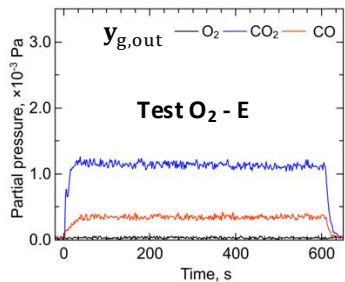
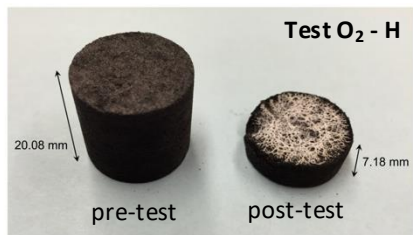
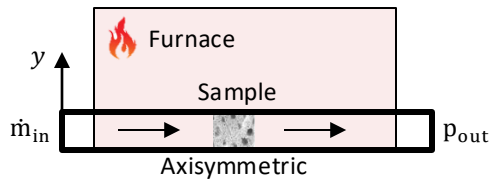


4D-MCT sample courtesy of Ringel^[20]

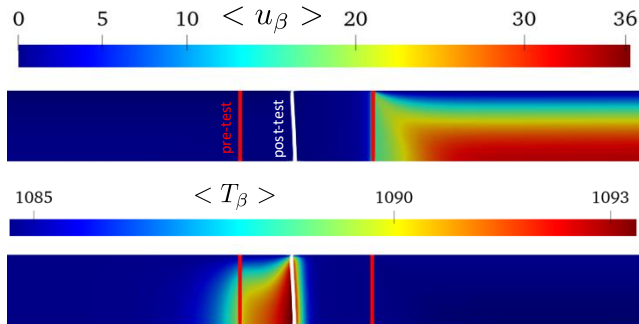


FiberForm surface area using filtering

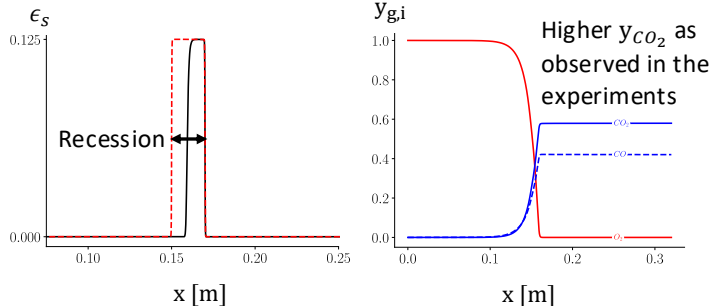
High-Temperature FiberForm Oxidation Flow Tube Reactor^[21]



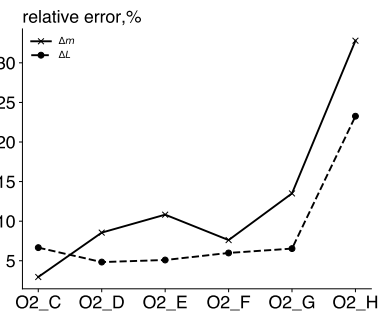
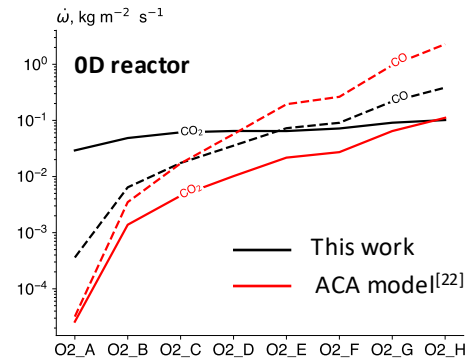
Combined CFD and Material Response PATO unified solver^[18]



Test O₂ - E



Oxidation Rates and Mass Loss Comparison to experiments



[18] Dias, Bruno, et al. "Numerical Simulation of FiberForm Using a Unified Flow-Material Approach: A Comparison With Flow-Tube Reactor Experiments." 2024.

[21] Panerai, Francesco, et al. "Experimental measurements of the high-temperature oxidation of carbon fibers." 2019.

[22] Prata, Krishna Sandeep, et al. "Air-carbon ablation model for hypersonic flight from molecular-beam data." 2022.



Conclusion

Flow and Material Coupling Strategies for Entry Systems

Coupling strategies play a crucial role in capturing multiphysics interactions between flow and material, providing varying levels of accuracy and computational efficiency depending on the application.

1. **One-Way Coupling** offers a computationally efficient approach for providing initial estimates in multiphysics problems. It is ideal for mission concept studies and the development of novel physics-based models (e.g., AERACEPT).
2. **Two-Way Coupling** ensures accurate modeling by capturing interactions between material mesh motion and flow fields through iterative exchanges. This approach is essential for arc-jet simulations (e.g., 3MDCP validation) and flight mission predictions involving trajectory and radiation (e.g., Dragonfly). Fidelity can also be enhanced through pyrolysis gas coupling.
3. **Unified Coupling** offers a combined flow and material solution within a single solver, eliminating the need for mesh motion and iterative steps. The oxidation model has been validated using flow tube experiments. It relies on a pressure-based solver, restricted to subsonic regimes. Future work will involve a coupling between the unified solver and hypersonic CFD.

Achieving the right balance between computational cost and fidelity is essential, as different applications demand different levels of precision to meet their specific goals.



References

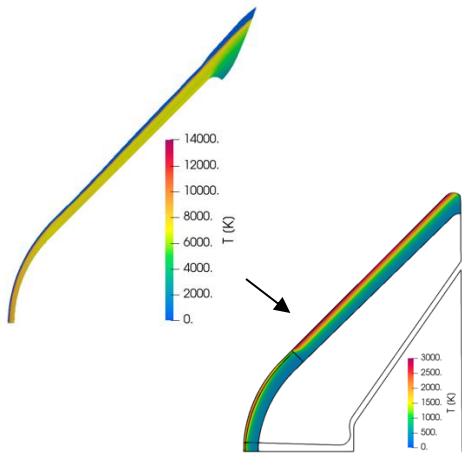
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- [8] Schroeder, Olivia, et al. "Ares: A Coupling Methodology for Ablation Modeling." AIAA SCITECH 2024 Forum. 2024.
- [9] Schulz, Joseph C., et al. "Development of a three-dimensional, unstructured material response design tool." 55th AIAA aerospace sciences meeting . 2017.
- [10] Rendall, Thomas CS, and Christian B. Allen. "Efficient mesh motion using radial basis functions with data reduction algorithms." Journal of Computational Physics 228.17 (2009): 6231-6249.
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- [12] Sahai, Amal, and Christopher O. Johnston. "On Computationally Efficient Radiative Transfer Calculations for Three-dimensional Entry Problems." AIAA SCITECH 2023 Forum. 2023.
- [13] Thornton, John M., et al. "Coupling heatsield response and aerothermal environment for mars entry via surface gas blowing." AIAA SCITECH 2023 Forum. 2023.
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- [16] Cross, Peter G., and Iain D. Boyd. "Conjugate analyses of ablation in rocket nozzles." Journal of Spacecraft and Rockets 56.5 (2019): 1593-1610.
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- [19] Panerai, Francesco, et al. "Experimental and numerical study of carbon fiber oxidation." Proc. 52nd Aerospace Sciences Meeting, AIAA paper. No. 2014-1208. 2014.
- [20] Ringel et al. "Carbon Fiber Oxidation" in 4D, In preparation.
- [21] Panerai, Francesco, et al. "Experimental measurements of the high-temperature oxidation of carbon fibers." International Journal of Heat and Mass Transfer 136 (2019): 972-986.
- [22] Prata, Krishna Sandeep, Thomas E. Schwartzentruber, and Timothy K. Minton. "Air-carbon ablation model for hypersonic flight from molecular-beam data." AIAA journal 60.2 (2022): 627-640.



Thank you for your attention! Any questions?

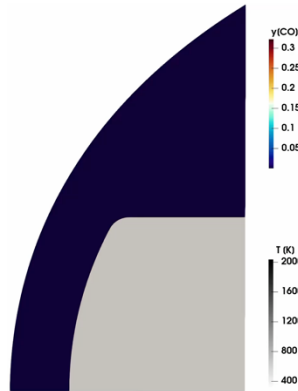
One-Way Coupling

Computationally efficient,
ideal for initial estimates



Two-Way Coupling

Accurate CFD-MR modeling with mesh motion,
computationally expensive due to iterations



Unified Coupling

Accurate flow-mat. modeling without mesh motion,
computationally efficient, under development

